

CBE 30357

9/7/17

TOPICS FOR TODAY

- 1) QUICK REVIEW OF DERIVATION OF MOMENTUM EQS
- 2) ... THEN TRANSFORMATION TO NAVIER-STOKES EQ'S
- 3) USE OF THESE TO SOLVE PROBLEMS
- 4) A LITTLE MORE ABOUT VISCOSITY

REVIEW: DERIVATION...

RATE OF CHANGE OF MOMENTUM
IN C.V.

$$\frac{d}{dt} \int \rho \vec{v} \Delta x \Delta y \Delta z$$

= RATE OF MOMENTUM FLOW
INTO C.V.

(E.G.) $\rightarrow \rho \vec{v} v_x \Delta y \Delta z \Big|_x$

- RATE OF MOMENTUM FLOW
OUT OF C.V.

$$- \rho \vec{v} v_x \Delta y \Delta z \Big|_{x+\Delta x}$$

+ Σ FORCES

BODY $\rightarrow \rho g \Delta x \Delta y \Delta z$ AREA

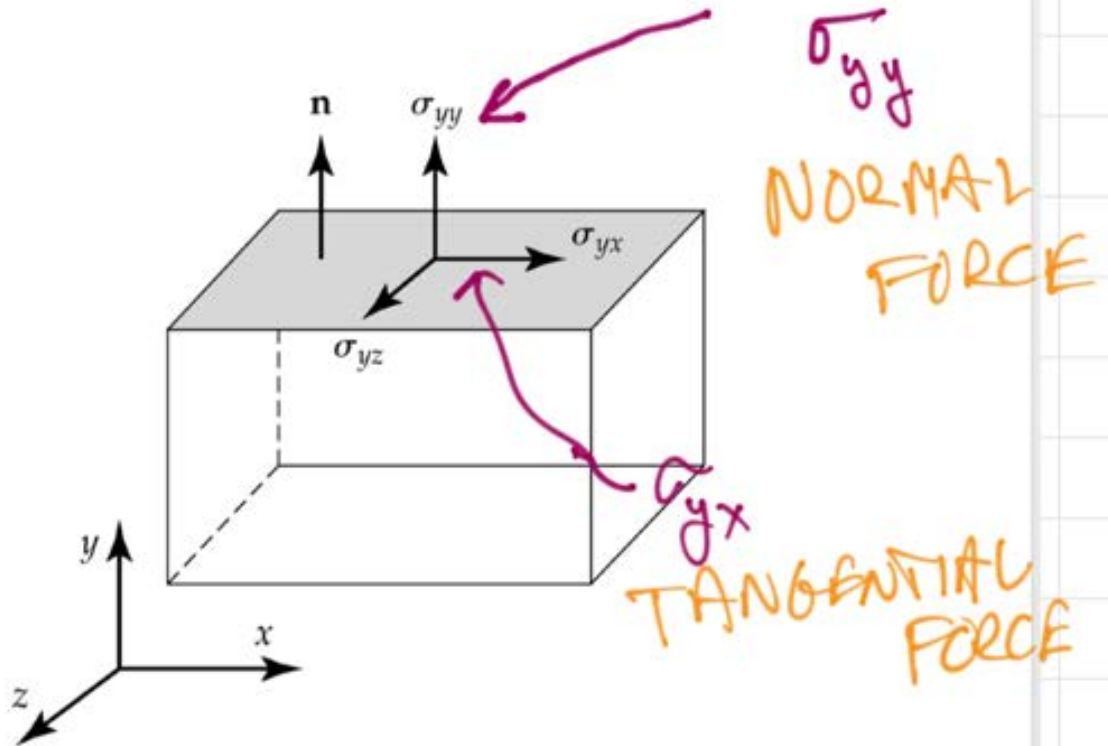
SURFACE F.C. $\rightarrow \sigma_{xx} (\Delta y \Delta z) + \sigma_{yx} \Delta z \Delta x + \sigma_{zx} \Delta x \Delta y$

FORCE/AREA $\begin{matrix} x \\ y \\ z \end{matrix}$

REVIEW

SURFACE FORCES

Figure 2.9 Cubic fluid element showing the stresses acting on a face of constant y . As shown, the normal stress σ_{yy} is tensile.



PEARSON

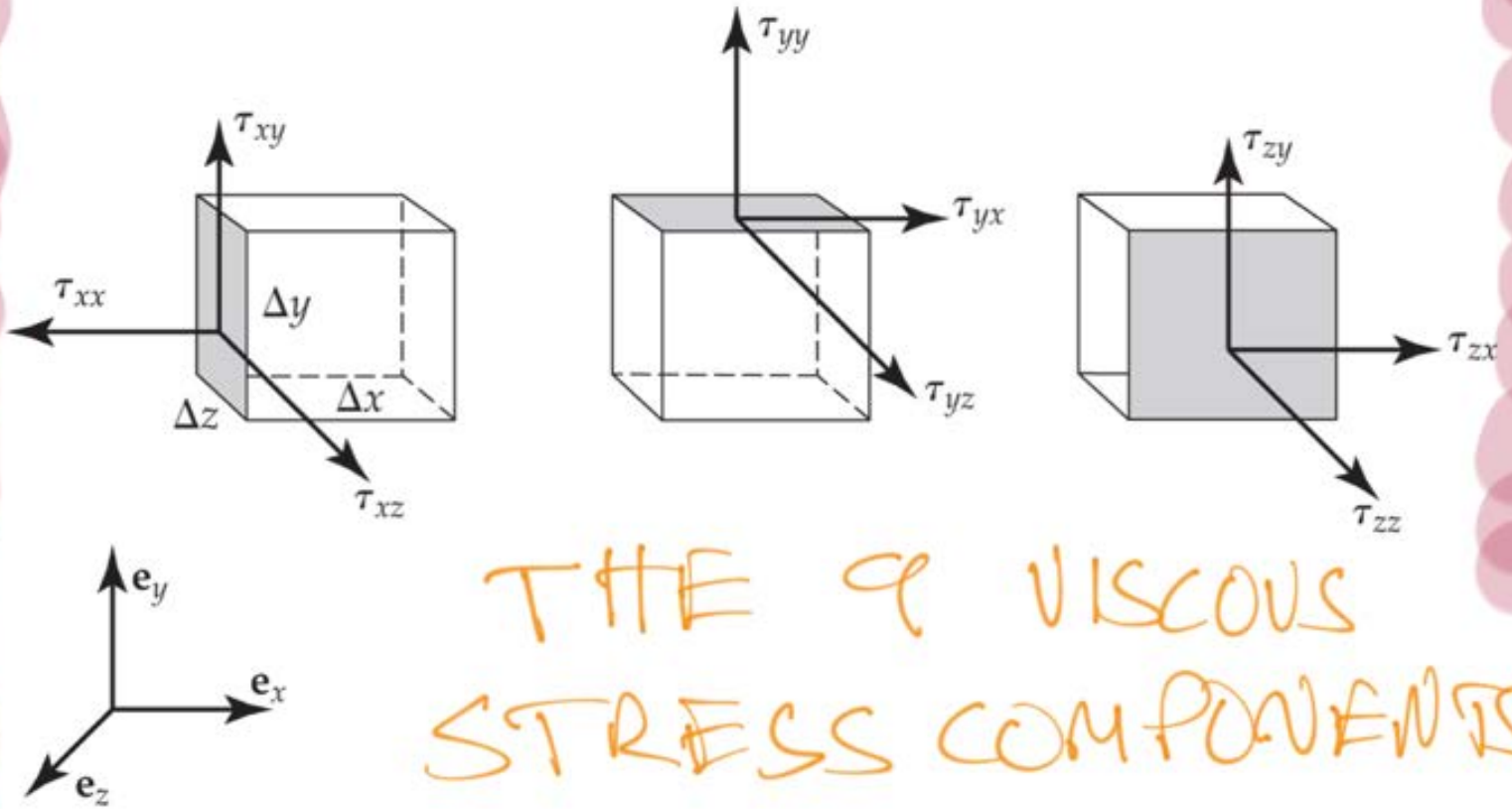
Transport Phenomena in Biological Systems, Second Edition
George A. Truskey, Fan Yuan, and David F. Katz

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σ_{yy}
FACE
(ORIENTATION)
DIRECTION
(OF FORCE)

REVIEW

Figure 3.5 Schematic of components of the viscous stress tensor.



THE 9 VISCOUS STRESS COMPONENTS

$$\vec{\sigma} = \vec{\tau} - \vec{I}p$$

REVIEW

COMPLETE EXPRESSION FOR SURFACE FORCES

$$\vec{Q} = \vec{\tau} - \vec{i} p \leftarrow \text{PRESSURE}$$

ALL SURFACE FORCES

VISCOUS SURFACE FORCES

$$\begin{cases} i = 1, 2, 3 \\ j = 1, 2, 3 \end{cases}$$

$$\vec{Q} = \tau_{ij} = \begin{pmatrix} \tau_{xx} - p & \tau_{yx} & \tau_{zx} \\ \tau_{xy} & \tau_{yy} - p & \tau_{zy} \\ \tau_{xz} & \tau_{yz} & \tau_{zz} - p \end{pmatrix}$$

$$\begin{cases} x \rightarrow 1 \\ y \rightarrow 2 \\ z \rightarrow 3 \end{cases}$$

TANGENTIAL COMPONENTS

NORMAL COMPONENTS

REVIEW

$$dV \left(\rho \frac{d\vec{v}}{dt} + \rho \vec{v} \cdot \nabla \vec{v} \right) \equiv$$

$$dV \left(-\nabla p + \nabla \cdot \vec{\tau} + \rho \vec{g} \right)$$

WHICH GIVES ...

$$\rho \frac{d\vec{v}}{dt} + \rho \vec{v} \cdot \nabla \vec{v} = -\nabla p + \nabla \cdot \vec{\tau} + \rho \vec{g}$$

CAUCHY - MOMENTUM EQS

Conservation of Linear Momentum

Rectangular coordinates

x component

$$\rho \left[\frac{\partial v_x}{\partial t} + v_x \frac{\partial v_x}{\partial x} + v_y \frac{\partial v_x}{\partial y} + v_z \frac{\partial v_x}{\partial z} \right] = \rho g_x - \frac{\partial p}{\partial x} + \left[\frac{\partial \tau_{xx}}{\partial x} + \frac{\partial \tau_{yx}}{\partial y} + \frac{\partial \tau_{zx}}{\partial z} \right]$$

y component

$$\rho \left[\frac{\partial v_y}{\partial t} + v_x \frac{\partial v_y}{\partial x} + v_y \frac{\partial v_y}{\partial y} + v_z \frac{\partial v_y}{\partial z} \right] = \rho g_y - \frac{\partial p}{\partial y} + \left[\frac{\partial \tau_{xy}}{\partial x} + \frac{\partial \tau_{yy}}{\partial y} + \frac{\partial \tau_{zy}}{\partial z} \right]$$

z component

$$\rho \left[\frac{\partial v_z}{\partial t} + v_x \frac{\partial v_z}{\partial x} + v_y \frac{\partial v_z}{\partial y} + v_z \frac{\partial v_z}{\partial z} \right] = \rho g_z - \frac{\partial p}{\partial z} + \left[\frac{\partial \tau_{xz}}{\partial x} + \frac{\partial \tau_{yz}}{\partial y} + \frac{\partial \tau_{zz}}{\partial z} \right]$$

"CAUCHY-MOMENTUM EQUATIONS"

THE ARE GENERALLY APPLICABLE

BUT THERE ARE TOO MANY
UNKNOWN

NEED CONSTITUTIVE EQUATION
TO RELATE τ 'S TO $\bar{\nabla} \bar{v}$

E.G. $\tau_{yx} = \mu \left(\frac{\partial v_x}{\partial y} + \frac{\partial v_y}{\partial x} \right)$

REVIEW

SIMPLEST VALID RELATION
IS CALLED: "NEWTONIAN
FLUID"

$$\bar{i} = 1, 2, 3$$

$$j = 1, 2, 3$$

$$\tau_{ij} = \mu \left(\frac{\partial v_i}{\partial x_j} + \frac{\partial v_j}{\partial x_i} \right)$$

SYMMETRY

$$\tau_{ji} = \tau_{ij}$$

ALWAYS

$$\mu > 0$$

FROM "2ND LAW"

Shear-Stress Tensor for an Incompressible Newtonian Fluid

Rectangular coordinates

$$\tau_{xx} = 2\mu \frac{\partial v_x}{\partial x} \quad (3.3.22a)$$

$$\tau_{yx} = \tau_{xy} = \mu \left(\frac{\partial v_x}{\partial y} + \frac{\partial v_y}{\partial x} \right) \quad (3.3.22b)$$

$$\tau_{zx} = \tau_{xz} = \mu \left(\frac{\partial v_x}{\partial z} + \frac{\partial v_z}{\partial x} \right) \quad (3.3.22c)$$

$$\tau_{yy} = 2\mu \frac{\partial v_y}{\partial y} \quad (3.3.22d)$$

$$\tau_{zy} = \tau_{yz} = \mu \left(\frac{\partial v_y}{\partial z} + \frac{\partial v_z}{\partial y} \right) \quad (3.3.22e)$$

$$\tau_{zz} = 2\mu \frac{\partial v_z}{\partial z} \quad (3.3.22f)$$

TABLE 3.4

Navier–Stokes Equation for an Incompressible Fluid

Rectangular coordinates

$$\frac{d\bar{P}}{dt} = \sum \bar{F}_i$$

x direction

$$\rho \left(\frac{\partial v_x}{\partial t} + v_x \frac{\partial v_x}{\partial x} + v_y \frac{\partial v_x}{\partial y} + v_z \frac{\partial v_x}{\partial z} \right) = -\frac{\partial p}{\partial x} + \mu \left[\frac{\partial^2 v_x}{\partial x^2} + \frac{\partial^2 v_x}{\partial y^2} + \frac{\partial^2 v_x}{\partial z^2} \right] + \rho g_x$$

y direction

$$\rho \left(\frac{\partial v_y}{\partial t} + v_x \frac{\partial v_y}{\partial x} + v_y \frac{\partial v_y}{\partial y} + v_z \frac{\partial v_y}{\partial z} \right) = -\frac{\partial p}{\partial y} + \mu \left[\frac{\partial^2 v_y}{\partial x^2} + \frac{\partial^2 v_y}{\partial y^2} + \frac{\partial^2 v_y}{\partial z^2} \right] + \rho g_y$$

z direction

$$\rho \left(\frac{\partial v_z}{\partial t} + v_x \frac{\partial v_z}{\partial x} + v_y \frac{\partial v_z}{\partial y} + v_z \frac{\partial v_z}{\partial z} \right) = -\frac{\partial p}{\partial z} + \mu \left[\frac{\partial^2 v_z}{\partial x^2} + \frac{\partial^2 v_z}{\partial y^2} + \frac{\partial^2 v_z}{\partial z^2} \right] + \rho g_z$$

TABLE 3.1

The Conservation of Mass (Continuity Equation)

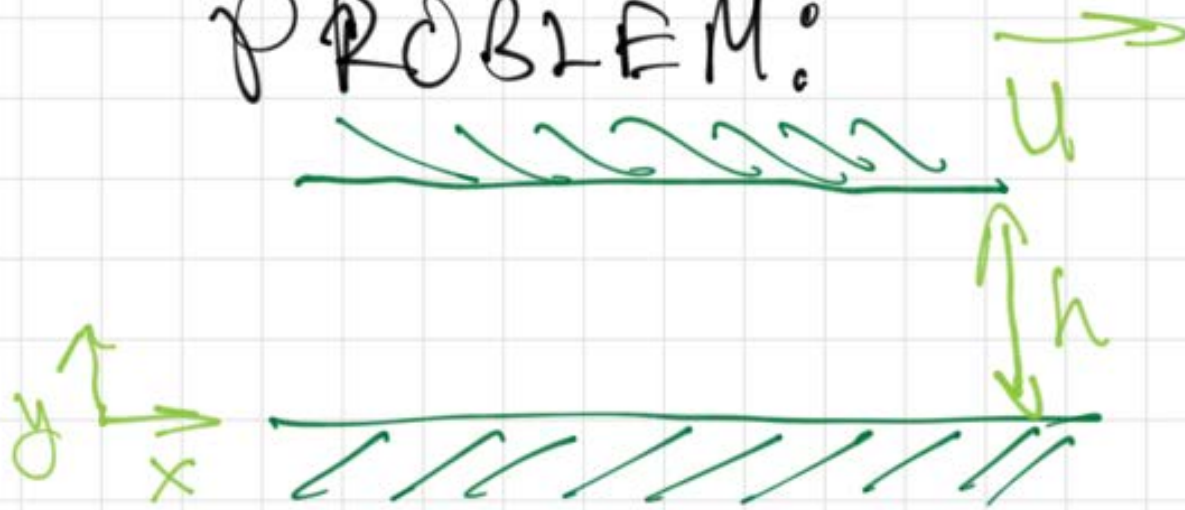
Rectangular coordinates (x, y, z)

$$\frac{\partial \rho}{\partial t} = - \left(\frac{\partial \rho v_x}{\partial x} + \frac{\partial \rho v_y}{\partial y} + \frac{\partial \rho v_z}{\partial z} \right)$$

These equations can be used to solve almost any fluid flow problem if the fluid is Newtonian.

Our challenge is to learn to use them!

WE START WITH
THE SIMPLEST
PROBLEM:



BOTTOM PLATE IS FIXED
TOP PLATE MOVES AT u

WHAT IS $v_x(y)$?

WHAT FORCE IS REQUIRED?

NEWTONIAN FLUID, μ

$$\rho = \text{CONST}$$

LOOK AT CONTINUITY

TABLE 3.1

The Conservation of Mass (Continuity Equation)

Rectangular coordinates (x, y, z)
$$\frac{\partial \rho}{\partial t} = -\left(\frac{\partial \rho v_x}{\partial x} + \frac{\partial \rho v_y}{\partial y} + \frac{\partial \rho v_z}{\partial z}\right)$$

$$\frac{\partial v_x}{\partial x} + \frac{\partial v_y}{\partial y} + \frac{\partial v_z}{\partial z} = 0$$

BUT $\frac{\partial}{\partial z} = 0$

NOTHING CHANGES W/ X

$$\frac{\partial}{\partial x} = 0 \quad \therefore \frac{\partial v_x}{\partial y} = 0$$

ONLY $v_x(y)$

TABLE 3.4

Navier-Stokes Equation for an Incompressible Fluid

Rectangular coordinates

x direction

$$\rho \left(\frac{\partial v_x}{\partial t} + v_x \frac{\partial v_x}{\partial x} + v_y \frac{\partial v_x}{\partial y} + v_z \frac{\partial v_x}{\partial z} \right) = -\frac{\partial p}{\partial x} + \mu \left[\frac{\partial^2 v_x}{\partial x^2} + \frac{\partial^2 v_x}{\partial y^2} + \frac{\partial^2 v_x}{\partial z^2} \right] + \rho g_x$$

y direction

$$\rho \left(\frac{\partial v_y}{\partial t} + v_x \frac{\partial v_y}{\partial x} + v_y \frac{\partial v_y}{\partial y} + v_z \frac{\partial v_y}{\partial z} \right) = -\frac{\partial p}{\partial y} + \mu \left[\frac{\partial^2 v_y}{\partial x^2} + \frac{\partial^2 v_y}{\partial y^2} + \frac{\partial^2 v_y}{\partial z^2} \right] + \rho g_y$$

z direction

$$\rho \left(\frac{\partial v_z}{\partial t} + v_x \frac{\partial v_z}{\partial x} + v_y \frac{\partial v_z}{\partial y} + v_z \frac{\partial v_z}{\partial z} \right) = -\frac{\partial p}{\partial z} + \mu \left[\frac{\partial^2 v_z}{\partial x^2} + \frac{\partial^2 v_z}{\partial y^2} + \frac{\partial^2 v_z}{\partial z^2} \right] + \rho g_z$$

AFTER SOME WORK

$$0 = \mu \frac{\partial^2 v_x}{\partial y^2}$$

BOUNDARY
CONDITIONS..

$$v_x(0) = 0, \quad v_x(h) = U$$

FLUID STICKS TO SOLID SURFACES!

SOLVE

$$0 = M \frac{d}{dy} \left(\frac{dv_x}{dy} \right)$$

$$\int d \left(\frac{dv_x}{dy} \right) = \int 0$$

$$\frac{dv_x}{dy} - C_1 = 0$$

$$\int d(v_x) = \int C_1 dy$$

$$v_x = C_1 y + C_2$$

FIT B.C.'S

$$V_x(0) = 0 \quad \therefore C_2 = 0$$

$$V_x(h) = U = C_1 h$$

$$\therefore C_1 = \frac{U}{h}$$

$$\therefore V_x(y) = \frac{U}{h} y$$

LINEAR PROFILE
 μ DIVIDED OUT
AND SO DOES NOT
AFFECT PROFILE.

WHICH STRESS COMPONENTS ARE IMPORTANT ??

Shear-Stress Tensor for an Incompressible Newtonian Fluid

Rectangular coordinates

$$\tau_{xx} = 2\mu \frac{\partial v_x}{\partial x} \quad (3.3.22a)$$

$$\tau_{yx} = \tau_{xy} = \mu \left(\frac{\partial v_x}{\partial y} + \frac{\partial v_y}{\partial x} \right) \quad (3.3.22b)$$

$$\tau_{zx} = \tau_{xz} = \mu \left(\frac{\partial v_x}{\partial z} + \frac{\partial v_z}{\partial x} \right) \quad (3.3.22c)$$

$$\tau_{yy} = 2\mu \frac{\partial v_y}{\partial y} \quad (3.3.22d)$$

$$\tau_{zy} = \tau_{yz} = \mu \left(\frac{\partial v_y}{\partial z} + \frac{\partial v_z}{\partial y} \right) \quad (3.3.22e)$$

$$\tau_{zz} = 2\mu \frac{\partial v_z}{\partial z} \quad (3.3.22f)$$

$$\tau_{yx} = \mu \frac{\partial v_x}{\partial y}$$

$v_x(y) = \frac{y}{h} U$ ← NO VISCOSITY
 CONSTANT, NOT A FUNCTION OF y
 $\tau_{yx} = \mu \frac{\partial v_x}{\partial y} = \frac{\mu U}{h}$ ← VISCOSITY IN STRESS

If the question is: how much force will it take to push the plate at a given velocity ?

You want: force/area * area

$$\text{FORCE } F_x = \tau_{yx} \times W \times L$$

(FORCE/AREA)

τ_{yx} W L

WIDTH IN LENGTH IN

$$= \mu \frac{u}{h} W L$$

μ $\frac{u}{h}$ W L

MORE VISCOUS $\uparrow F$ FASTER $\uparrow F$ CLOSER SPACING $\uparrow F$

It is obvious, but what is the average velocity of the fluid?

$$\text{AVERAGE VELOCITY} = \frac{\text{TOTAL VOLUMETRIC FLOW}}{\text{AREA OF FLOW}}$$

$$\langle v_x \rangle = \frac{\int_0^w \int_0^h v_x(y) dy dz}{\int_0^w \int_0^h dy dz}$$

$$= \frac{\int_0^w dz \int_0^h \frac{uy}{h} dy}{wh}$$

$$\langle v_x \rangle = \frac{w \frac{u}{h} \left. \frac{y^2}{2} \right|_0^h}{wh} = \frac{u}{2}$$

SUPPOSE WE KEEP THE FLOW GEOMETRY THE SAME, BUT CAUSE THE FLOW BY A PRESSURE GRADIENT

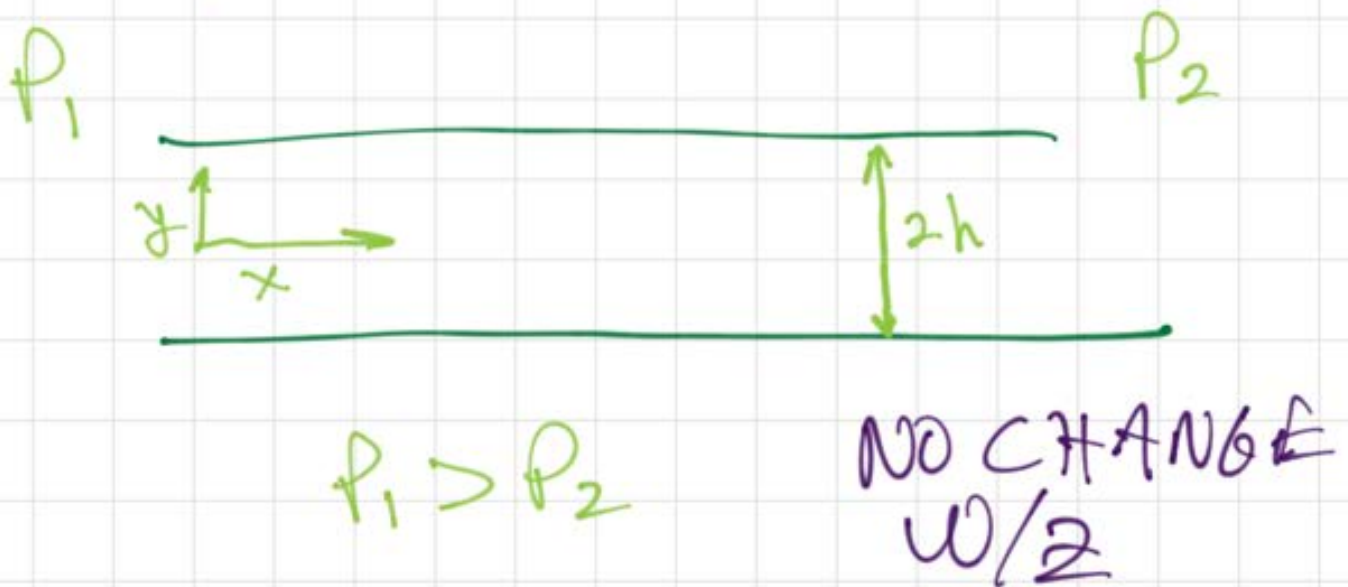


TABLE 3.4

Navier-Stokes Equation for an Incompressible Fluid

Rectangular coordinates

x direction

$$\rho \left(\frac{\partial v_x}{\partial t} + v_x \frac{\partial v_x}{\partial x} + v_y \frac{\partial v_x}{\partial y} + v_z \frac{\partial v_x}{\partial z} \right) = -\frac{\partial p}{\partial x} + \mu \left[\frac{\partial^2 v_x}{\partial x^2} + \frac{\partial^2 v_x}{\partial y^2} + \frac{\partial^2 v_x}{\partial z^2} \right] + \rho g_x$$

y direction

$$\rho \left(\frac{\partial v_y}{\partial t} + v_x \frac{\partial v_y}{\partial x} + v_y \frac{\partial v_y}{\partial y} + v_z \frac{\partial v_y}{\partial z} \right) = -\frac{\partial p}{\partial y} + \mu \left[\frac{\partial^2 v_y}{\partial x^2} + \frac{\partial^2 v_y}{\partial y^2} + \frac{\partial^2 v_y}{\partial z^2} \right] + \rho g_y$$

z direction

$$\rho \left(\frac{\partial v_z}{\partial t} + v_x \frac{\partial v_z}{\partial x} + v_y \frac{\partial v_z}{\partial y} + v_z \frac{\partial v_z}{\partial z} \right) = -\frac{\partial p}{\partial z} + \mu \left[\frac{\partial^2 v_z}{\partial x^2} + \frac{\partial^2 v_z}{\partial y^2} + \frac{\partial^2 v_z}{\partial z^2} \right] + \rho g_z$$

$$0 = -\frac{\partial p}{\partial x} + \mu \frac{\partial^2 v_x}{\partial y^2}$$

BOUNDARY CONDITIONS:

$$v_x(h) = 0$$

$$v_x(-h) = 0$$

[NO
SLIP]

WE SOLVE:

$$\frac{d^2 v_x}{dy^2} = \frac{1}{\mu} \frac{dp}{dx}$$

A subtle and perhaps confusing argument will now be given! 🤔

We conclude that v_x does not change in x .

We conclude that $\frac{\partial p}{\partial x}$ does not change with y

Hence, the only way for these terms to be always equal is if they are equal to a constant.

$$\frac{d^2 v_x}{dy^2} = \frac{1}{\mu} \frac{dp}{dx} = \text{CONSTANT}$$

NOW WE PROCEED...

$$\frac{d}{dy} \left(\frac{dv_x}{dy} \right) = \frac{1}{\mu} \frac{dp}{dx}$$

$$d \left(\frac{dv_x}{dy} \right) = \frac{1}{\mu} \frac{dp}{dx} dy$$

$$\int d \left(\frac{dv_x}{dy} \right) = \int \frac{1}{\mu} \frac{dp}{dx} dy$$

$$\frac{dv_x}{dy} = \frac{1}{\mu} \frac{dP}{dx} y + C_1$$

$$dv_x = \left(\frac{1}{\mu} \frac{dP}{dx} y + C_1 \right) dy$$

$$\int dv_x = \int \left(\frac{1}{\mu} \frac{dP}{dx} y + C_1 \right) dy$$

$$v_x = \frac{1}{\mu} \frac{dP}{dx} \frac{y^2}{2} + C_1 y + C_2$$

NOW WE NEED TO APPLY
BOUNDARY CONDITIONS

$$v_x(h) = 0 = \frac{1}{\mu} \frac{dP}{dx} \frac{h^2}{2} + C_1 h + C_2$$

$$v_x(-h) = 0 = \frac{1}{\mu} \frac{dP}{dx} \frac{h^2}{2} - C_1 h + C_2$$

$$\therefore C_1 = 0$$

$$C_2 = -\frac{1}{\mu} \frac{dP}{dx} \frac{h^2}{2}$$

WHICH GIVES

$$v_x(y) = -\frac{1}{\mu} \frac{dP}{dx} \frac{h^2}{2} \left(1 - \frac{y^2}{h^2} \right)$$

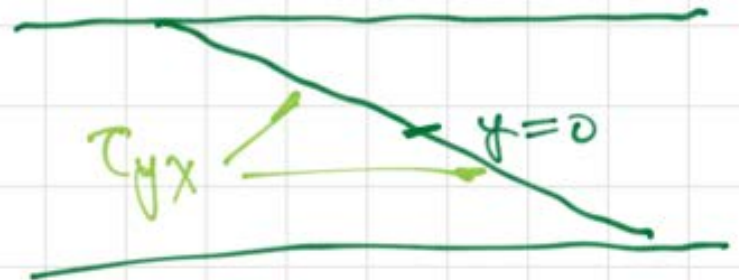


SHEAR STRESS

$$\tau_{yx} = \mu \frac{\partial v_x}{\partial y}$$

$$= \frac{dP}{dx} y$$

(-)



STRESS IS MAXIMUM AT WALLS, 0 IN MIDDLE

WHAT IS AVERAGE VELOCITY?

$$\langle v_x \rangle = \frac{\int_0^w \int_{-h}^h v_x(y) dy dz}{\int_0^w \int_{-h}^h dy dz}$$

$$v_x(y) = -\frac{1}{\mu} \frac{dp}{dx} \frac{h^2}{2} \left(1 - \frac{y^2}{h^2} \right)$$

$$\langle v_x \rangle = \frac{-\frac{1}{\mu} \frac{dp}{dx} \frac{h^2}{2} w}{2wh} \int_{-h}^h \left(1 - \frac{y^2}{h^2} \right) dy$$

$$= \frac{-\frac{1}{\mu} \frac{dp}{dx} \frac{h^2 w}{2}}{2w} \left(y - \frac{y^3}{3h^2} \right) \Big|_{-h}^h$$

$$= \frac{-1}{4\mu} \frac{dp}{dx} h \left(\left(h - \frac{1}{3}h \right) - \left(-h + \frac{1}{3}h \right) \right)$$

$$\langle u_x \rangle = \frac{1}{3\mu} \left(-\frac{dp}{dx} \right) h^2$$

PRESSURE "DROPS" LINEARLY WITH DISTANCE, HENCE

$$\frac{dp}{dx} < 0$$

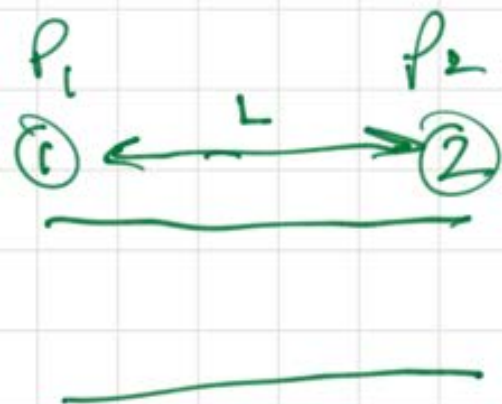
$$\frac{d^2 u_x}{dy^2} = \frac{1}{\mu} \frac{dp}{dx} = \text{CONSTANT}$$

$$\frac{dp}{dx} = C_1$$

$$dp = C_1 dx$$

$$\int dp = \int C_1 dx$$

$$p = C_1 x + C_2$$



$$\textcircled{1} x=0, P=P_1$$

$$\therefore P = C_1 x + P_1$$

$$\text{IF } P = P_2 \text{ @ } x = L$$

$$P_2 = C_1 L + P_1$$

$$C_1 = \frac{P_2 - P_1}{L}$$

OR

$$P = \left[\frac{P_2 - P_1}{L} \right] x + P_1$$
$$= \frac{dP}{dx} L \textcircled{1}$$

PRESSURE DECREASES
LINEARLY WITH DISTANCE

How does gravity cause fluid flow?



X-DIRECTION NAVIER-STOKES EQ.

$$0 = \mu \frac{d^2 v_x}{dy^2} + \rho g \cos \theta$$

BOUNDARY CONDITIONS

$$v_x(0) = 0$$

AT $y = h$ WE HAVE TO FIND
SOMETHING ELSE.....

$$\tau_{yx}(y=h) = 0 \quad \text{NO STRESS} \quad \text{📞}$$

Boundary conditions:

1. Fluid sticks to solid surfaces

$$v_x(0) = 0$$

2. Fluid sticks to another immiscible fluid

$$v_x^I(h) = v_x^{II}(h)$$

3. The shear stress is continuous across the fluid-fluid interface

$$\tau_{yx}^I(h) = \tau_{yx}^{II}(h)$$
$$\mu^I \frac{\partial v_x^I}{\partial y} \Big|_h = \mu^{II} \frac{\partial v_x^{II}}{\partial y}$$

4. For liquid flowing next to quiescent air, the stress can be considered approximately 0.

$$\tau_{yx}(h) \approx 0$$

$$\frac{\partial v_x}{\partial y}(h) \approx 0$$

"FALLING"
FILM

$$0 = \mu \frac{d^2 v_x}{dy^2} + \rho g \cos \theta$$

$$\frac{d^2 v_x}{dy^2} = -\frac{\rho g}{\mu} \cos \theta$$

$$\frac{d}{dy} \left(\frac{dv_x}{dy} \right) = -\frac{\rho g}{\mu} \cos \theta$$

$$\int d \left(\frac{dv_x}{dy} \right) = -\frac{\rho g}{\mu} \cos \theta \int dy$$

$$\frac{dv_x}{dy} = -\frac{\rho g}{\mu} \cos \theta y + C_1$$

$$\int dv_x = \int \left(-\frac{\rho g}{\mu} \cos \theta y + C_1 \right) dy$$

$$v_x = -\frac{\rho g}{\mu} \cos \theta \frac{y^2}{2} + C_1 y + C_2$$

$$\textcircled{1} \quad y=0 \quad v_x(0)=0$$

$$\therefore C_2 = 0$$

$$\textcircled{2} \quad y=h \quad \frac{dv_x}{dy} = 0$$

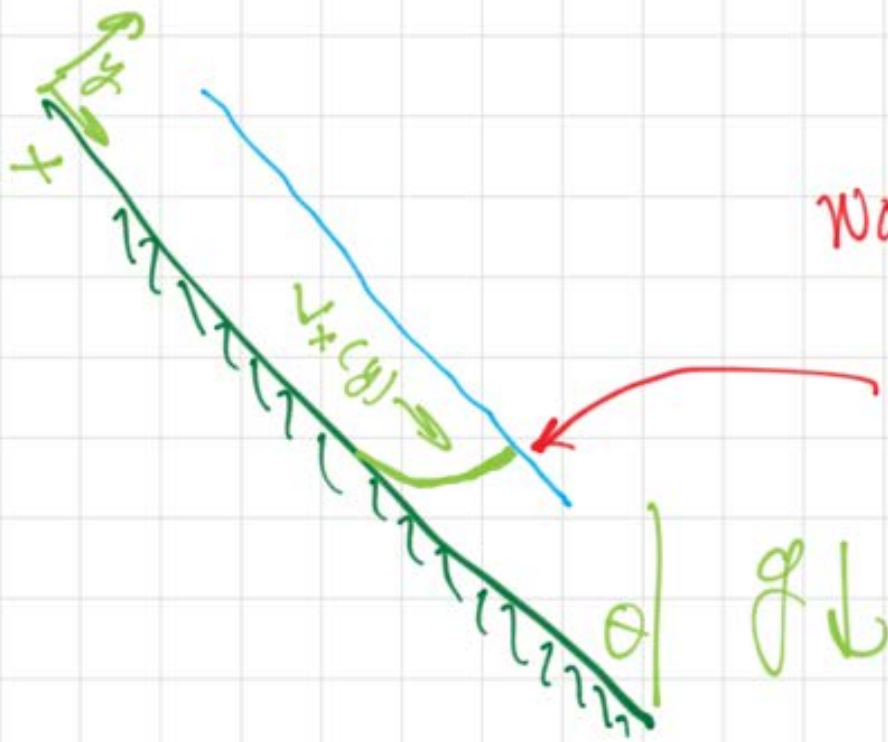
$$\frac{dv_x}{dy} = -\frac{\rho g}{\mu} \cos \theta y + C_1$$

$$\therefore C_1 = \frac{\rho g}{\mu} \cos \theta h$$

THUS THE PROFILE IS

$$v_x(y) = \frac{\rho g}{\mu} \cos \theta \left(\frac{y^2}{2} - yh \right)$$





NO STRESS

$$\frac{dv_x}{dy} \Big|_{y=0} = h$$